

SUPPLEMENTAL TYPE CERTIFICATE

10073903

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

SEATON ENGINEERING CORPORATION

40145 SUNSET DRIVE LOON LAKE 99148 UNITED STATES OF AMERICA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: SEE FAA APPROVED MODEL LIST

(AML) SA01861SE, DATED 19 JULY 2017.

Type Certificate Holder: See FAA AML

Type: See FAA AML Model: See FAA AML

Original STC Number: FAA STC SA01861SE

Description of Design Change:

Installation of Landing Light Controler

EASA Certification Basis:

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA01861SE.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

Seaton Engineering Corporation Master Drawing List as listed on AML SA01861SE.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 29 July 2020

Dominique ROLAND Head of Department General Aviation



Task Number: 60074537

SEATON ENGINEERING CORPORATION - 310462



Instructions for Continued Airworthiness (ICA), Document Number 9150-008, Revision D, dated 25 June 2013, or later FAA accepted revision.

FAA approved Aircraft Flight Manual Supplement as listed on AML SA01861SE.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

This STC is valid for aircraft types and models approved by EASA and listed on the FAA AML SA01861SE.

